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# SHOCK-MODELLING WITH BODY-FITTED MESHES USING A LOCAL PARAMETERISATION OF THE EQUATION OF STATE

by

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**Numerical Analysis Report 1/88** 

# DEPARTMENT OF MATHEMATICS

# SHOCK-MODELLING WITH BODY-FITTED MESHES USING A LOCAL PARAMETERISATION OF THE EQUATION OF STATE

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#### NUMERICAL ANALYSIS REPORT 1/88

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This work forms part of the research programme for the Institute of Computational Fluid Dynamics at the Universities of Oxford and Reading and was funded by AWRE, Aldermaston under Contract No. NSN/13B/2A88719

#### ABSTRACT

An efficient algorithm based on flux difference splitting is presented for the solution of the three-dimensional Euler equations of gas dynamics in a generalised coordinate system with a general equation of state. The scheme is based on solving linearised Riemann problems approximately and in more than one dimension incorporates operator splitting. The algorithm uses a local parameterisation of the equation of state and as a consequence requires only one function evaluation in each computational cell. The scheme has good shock capturing properties and the advantage of using body-fitted meshes. Numerical results are shown for Mach 8 flow of 'equilibrium air' past a circular cylinder.

#### KEYWORDS

Euler equations, Riemann solver, Body-fitted coordinates, Parameterisation of the equation of state.

#### 1. INTRODUCTION

In 1981 Roe [1] proposed an approximate (linearised) Riemann solver for the solution of the three-dimensional Euler equations in Cartesian coordinates with an ideal gas. A similar scheme was proposed by Glaister [2] for the Euler equations in Cartesian coordinates with general convex equations of state. A disadvantage of Glaister's scheme is that four function evaluations are required in each computational cell to approximate the first derivatives of the For complex equations of state, e.g. curve fits equation of state. for equilibrium air [3] this can prove to be an expensive overhead. We seek here to devise a scheme that requires only one function evaluation in each cell with no deterioration in the quality of the solution. This is achieved by a local parameterisation of the equation of state, in effect a 'variable effective gamma' (VEG) scheme. Furthermore, the proposed scheme applies to a generalised coordinate system, and when incorporated with operator splitting leads to an efficient algorithm that has good shock-capturing properties and the advantage of using body-fitted meshes.

In §2 we consider the Jacobian matrix of one of the flux functions for the Euler equations in a generalised coordinate system, and in §3 derive an approximate Riemann solver for the solution of these equations. Finally, in §4 we give the numerical results achieved for a two-dimensional test problem using the scheme of §3.

#### 2. EULER EQUATIONS

In this section we state the equations of motion for an inviscid, compressible fluid in three dimensions in terms of three generalised space coordinates. We also give the eigenvalues and eigenvectors of the Jacobian matrix of one of the corresponding flux functions.

#### 2.1 Equations of flow

The three-dimensional Euler equations for the flow of an inviscid, compressible fluid can be written in generalised coordinates  $\xi$ ,  $\eta$ ,  $\zeta$  as (see Appendix C)

$$(J_{\widetilde{W}})_{t} + \widetilde{E}_{F} + \widetilde{G}_{n} + \widetilde{H}_{C} = 0.$$
 (1)

where

$$\underline{\mathbf{w}} = \left[\rho, \rho \mathbf{u}, \rho \mathbf{v}, \rho \mathbf{w}, \mathbf{e}\right]^{\mathrm{T}}$$
(2)

$$\widetilde{E}(\underline{w}) = \left[\rho U, J_{\xi}^{\mathbf{x}} \mathbf{p} + \rho \mathbf{u} U, J_{\xi}^{\mathbf{y}} \mathbf{p} + \rho \mathbf{v} U, J_{\xi}^{\mathbf{z}} + \rho \mathbf{w} U, U(\mathbf{e} + \mathbf{p})\right]^{T}$$
(3)

$$\mathbf{G}(\mathbf{w}) = \left[\rho \mathbf{V}, \ \mathbf{J}_{\eta}^{\mathbf{x}} \mathbf{p} + \mathbf{p} \mathbf{u} \mathbf{V}, \ \mathbf{J}_{\eta}^{\mathbf{y}} \mathbf{p} + \rho \mathbf{v} \mathbf{V}, \ \mathbf{J}_{\eta}^{\mathbf{z}} \mathbf{p} + \rho \mathbf{w} \mathbf{V}, \ \mathbf{V}(\mathbf{e} + \mathbf{p})\right]^{\mathbf{T}}$$
(4)

$$\underbrace{\mathbb{H}(\underline{w})} = \left[\rho \mathbb{W}, \ J_{\zeta}^{x} \mathbf{p} + \rho \mathbf{u} \mathbb{W}, \ J_{\zeta}^{y} \mathbf{p} + \rho \mathbf{v} \mathbb{W}, \ J_{\zeta}^{z} \mathbf{p} + \rho \mathbf{w} \mathbb{W}, \ \mathbb{W}(\mathbf{e} + \mathbf{p})\right]^{T} \tag{5}$$

$$e = \rho i + \frac{1}{2}\rho(u^2 + v^2 + w^2)$$
 (6)

and

$$U = J_{\xi}^{x}u + J_{\xi}^{y}v + J_{\xi}^{z}w \qquad (7)$$

$$V = J_n^x u + J_n^y v + J_n^z w$$
 (8)

$$W = J_{\zeta}^{x}u + J_{\zeta}^{y}v + J_{\zeta}^{z}w . \qquad (9)$$

The Jacobian of the grid transformation  $x = x(\xi, \eta, \zeta)$ ,  $y = y(\xi, \eta, \zeta)$ ,  $z = z(\xi, \eta, \zeta)$  from Cartesian coordinates (x, y, z) to generalised coordinates  $(\xi, \eta, \zeta)$  is given by

$$J = \begin{vmatrix} x_{\xi} & y_{\xi} & z_{\xi} \\ x_{\eta} & y_{\eta} & z_{\eta} \\ x_{\zeta} & y_{\zeta} & z_{\zeta} \end{vmatrix}; \qquad (10)$$

whilst the cofactors of this matrix are represented by  $J_{\xi}^{x}$  etc. The cofactor of  $x_{\eta}$  is given by

$$J_{\xi}^{x} = y_{\eta} z_{\zeta} - z_{\eta} y_{\zeta} , \qquad (11)$$

with similar expressions for  $J_{\xi}^{y}$  etc. The quantities  $(\rho, u, v, w, p, i, e) = (\rho, u, v, w, p, i, e)(\xi, \eta, \zeta, t)$  represent the density, velocity in the x, y and z coordinate directions, pressure, specific internal energy and total energy, respectively, at a general position  $\xi, \eta, \zeta$  in space and at time t . In addition we have an equation of state of the form

$$p = p(\rho, i) . \qquad (12)$$

#### 2.2 Structure of the Jacobian matrix

We now give the Jacobian matrix of the flux function F(w), and its eigenvalues and eigenvectors, since this information, together with similar information for the Jacobian matrices of G(w) and H(w), will form the basis for the approximate Riemann solver.

The Jacobian matrix  $A=\frac{\partial \widetilde{F}}{\partial \underline{w}}$  of the flux function  $\widetilde{F}(\underline{w})$  is given by

$$A = \begin{bmatrix} 0 & J_{\xi}^{x} & J_{\xi}^{y} & J_{\xi}^{z} & 0 \\ RJ_{\xi}^{x} - uU & U + uSJ_{\xi}^{x} & uJ_{\xi}^{y} - vTJ_{\xi}^{x} & uJ_{\xi}^{z} - wTJ_{\xi}^{x} & TJ_{\xi}^{x} \\ RJ_{\xi}^{y} - vU & vJ_{\xi}^{x} - uTJ_{\xi}^{y} & U + vSJ_{\xi}^{y} & vJ_{\xi}^{z} - wTJ_{\xi}^{y} & TJ_{\xi}^{y} \\ RJ_{\xi}^{z} - wU & wJ_{\xi}^{x} - uTJ_{\xi}^{z} & wJ_{\xi}^{y} - vTJ_{\xi}^{z} & U + wSJ_{\xi}^{z} & TJ_{\xi}^{z} \\ U(R-H) & HJ_{\xi}^{x} - uUT & HJ_{\xi}^{y} - vUT & HJ_{\xi}^{z} - wUT & U + UT \end{bmatrix}$$
(13)

where for convenience we have written

$$R = a^2 - \frac{p_i}{\rho} (H - q^2) , \qquad (14)$$

$$S = 1 - \frac{p_i}{\rho} \tag{15}$$

$$T = \frac{p_i}{\rho} . ag{16}$$

The fluid speed q, enthalpy H and sound speed a are given by

$$q^2 = u^2 + v^2 + w^2 , (17)$$

$$H = \frac{p}{\rho} + i + \frac{1}{2}q^{2}, \qquad (18)$$

and

$$a^{2} = \frac{pp_{1}}{\rho^{2}} + p_{\rho} . {19}$$

and the quantities  $p_i$ ,  $p_\rho$  denote the derivatives  $\frac{\partial p}{\partial i}(\rho, i)$ 

 $\frac{\partial p}{\partial \rho}(\rho, i)\Big|_i$  , respectively. The eigenvalues of A are given by

$$\lambda_{1,2,3,4,5} = U \pm aD, U, U, U$$
 (20a-e)

with corresponding linearly independent eigenvectors

$$\underset{\sim}{e}_{1,2} = \left[1, u \pm \frac{aJ_{\xi}^{x}}{D}, v \pm \frac{aJ_{\xi}^{y}}{D}, w \pm \frac{aJ_{\xi}^{z}}{D}, H \pm \frac{aU}{D}\right]^{T}$$
 (21a-b)

$$e_3 = \left[1, u, v, w, \frac{1}{2}q^2 + i - \frac{\rho p_{\rho}}{p_i}\right]^T$$
 (21c)

$$e_4 = \left[0, -J_{\xi}^{y}, J_{\xi}^{x}, 0, vJ_{\xi}^{x} - uJ_{\xi}^{y}\right]^T$$
 (21d)

and

$$e_5 = \left[0, -J_{\xi}^{z}, 0, J_{\xi}^{x}, wJ_{\xi}^{x} - uJ_{\xi}^{z}\right]^{T}$$
 (21e)

where

$$D = \sqrt{\left[J_{\xi}^{x}\right]^{2} + \left[J_{\xi}^{y}\right]^{2} + \left[J_{\xi}^{z}\right]^{2}} . \qquad (22)$$

Similar results hold for the Jacobian matrices of G(w) and H(w).

In the next section we develop an approximate Riemann solver based on the results of this section.

#### 3. APPROXIMATE RIEMANN SOLVER

In this section we derive an approximate Riemann solver for the solution of equations (1)-(12).

We propose solving equations (1)-(12) using operator splitting, i.e. we solve successively

$$(J_{\widetilde{W}})_{t} + F_{F} = 0$$
 (23a)

$$(J_{\widetilde{W}})_{t} + G_{\eta} = O$$
 (23b)

$$(J_{\widetilde{W}})_{t} + H_{\widetilde{\zeta}} = 0$$
 (23c)

along  $\xi$ ,  $\eta$  and  $\zeta$  coordinate lines, respectively. We describe the scheme for solving equation (23a) and the solution of equations (23b) and (23c) will follow in a similar way.

#### 3.1 Parameterisation of the equation of state

The equation of state for an ideal gas is given by

$$p = (\gamma - 1)\rho i \tag{24}$$

where  $\gamma$  is a constant and represents the ratio of specific heat capacities of the fluid. Following this, for a general equation of state  $p = p(\rho, i)$  we define a new dependent variable  $\gamma = \gamma(\rho, i)$  by

$$\gamma(\rho, i) = \frac{p(\rho, i)}{\rho i} + 1 \tag{25}$$

so that the equation of state (12) can be rewritten as

$$p = \left[ \gamma(\rho, i) - 1 \right] \rho i . \qquad (26)$$

(Many equations of state for real gases are already given in the form of equation (26). The ideal equation of state is given by  $\gamma \equiv \text{constant.}$ )

From equation (25), the eigenvectors  $\stackrel{\rm e}{\sim}_{1,2}$  of equations (21a-b) can be rewritten in terms of  $\gamma$  as

$$e_{1,2} = \left[1, u \pm \frac{aJ_{\xi}^{x}}{D}, v \pm \frac{aJ_{\xi}^{y}}{D}, w \pm \frac{aJ_{\xi}^{z}}{D}, \frac{\gamma p}{(\gamma-1)\rho} + 2q^{2} \pm \frac{aU}{D}\right]^{T}. \quad (27a-b)$$

In particular, for the ideal equation of state (25) the sound speed a is given by equation (19) as

$$a^2 = \frac{\gamma p}{\rho} \quad , \tag{28}$$

and the fifth component of  $\stackrel{e}{\approx}_3$  , given by equation (21c), becomes  ${}^{1}\!\!/\!\! q^2$  since  $i-\frac{\rho p}{p_4}=0$  .

#### 3.2 Linearised Riemann Problem

If the solution of equation (23a) is sought along a  $\xi$  coordinate line given by  $\eta=\eta_0$ ,  $\zeta=\zeta_0$ , constants, using a finite difference method then the solution is known at a set of discrete mesh points  $(\xi,\,\eta,\,\zeta,\,t)=(\xi_j,\,\eta_0,\,\zeta_0,\,t_n)$  at any time  $t_n$ . Following Godunov [4] the approximate solution  $\psi_j^n$  to  $\psi_j$  at  $(\xi_j,\,\eta_0,\,\zeta_0,\,t_n)$  can be considered as a set of piecewise constants  $\psi_j=\psi_j^n$  for  $\xi\in\left[\xi_j-\frac{\Delta\xi}{2},\,\xi_j+\frac{\Delta\xi}{2}\right]$  at time  $t_n$ , where  $\Delta\xi=\xi_j-\xi_{j-1}$  is a constant mesh spacing. A Riemann problem is now present at each interface  $\xi_{j-1}=\psi(\xi_{j-1}+\xi_j)$  separating adjacent states  $\psi_{j-1}^n,\,\psi_j^n$ . We consider solving the linearised Riemann problem

$$(J_{\underline{w}})_{t} + \widetilde{A} \left[ \underbrace{w}_{j-1}^{n}, \underbrace{w}_{j}^{n} \right] \underbrace{w}_{\xi} = 0$$
 (29)

where  $\overset{\sim}{A}_{j-\frac{1}{2}}=\overset{\sim}{A}\begin{bmatrix} \overset{n}{w}_{j-1}, & \overset{n}{w}_{j} \end{bmatrix}$  is an approximation to the Jacobian matrix A

and is a constant matrix depending on the states either side of  $\xi_{j-1/2}$ . The matrix  $\overset{\sim}{A}_{j-1/2}$  will be required to satisfy the following properties

(i)  $\stackrel{\sim}{A}_{j-\frac{1}{2}}$  has five linearly independent eigenvectors

(ii) 
$$\Delta_{\widetilde{\Sigma}}^{F} = \widetilde{A}_{j-2} \Delta_{\widetilde{W}}^{W}.$$

These properties were shown by Roe [1] in the ideal gas case in Cartesian coordinates to guarantee conservation and have good one-dimensional shock-capturing properties.

#### 3.3 Numerical scheme

and

Once such a matrix has been constructed equation (29) can be solved approximately as

$$J_{j \to k} \frac{(w_k^{n+1} - w_k^n)}{\Delta t} + \widetilde{A}_{j \to k} \frac{(w_j^n - w_{j-1}^n)}{\Delta \xi} = 0$$
 (30)

where k can be j-1 or j ,  $\Delta t = t_{n+1} = t_n$  is a constant time step and  $J_{j-1/2}$  is an approximation to the grid Jacobian at  $(\xi, \eta, \zeta) = (\xi_{j-1/2}, \eta_0, \zeta_0)$  . If we project

$$\Delta \underline{w} = \underline{w}_{j}^{n} - \underline{w}_{j-1}^{n} = \sum_{i=1}^{5} \widetilde{\alpha}_{i} \widetilde{z}_{i}$$
 (31)

where  $\overset{\sim}{r_i}$  are the eigenvectors of  $\overset{\sim}{A_{j-1/2}}$  then equation (30) can be written as

$$J_{j-\frac{1}{2}} \frac{(\underline{w}_{k}^{n+1} - \underline{w}_{k}^{n})}{\Delta t} + \frac{\sum_{i=1}^{5} \widetilde{\lambda}_{i} \widetilde{\alpha}_{i} \widetilde{\Sigma}_{i}}{\Delta \xi} = 0$$
 (32)

where  $\tilde{\lambda}_i$  are the eigenvalues of  $\tilde{A}_{j-1/2}$ . Equation (32) now gives rise to the following first order upwind algorithm

$$\mathbf{w}_{j-1}^{n+1} = \mathbf{w}_{j-1}^{n} - \frac{\Delta t}{J_{j-2}\Delta \xi} \widetilde{\lambda}_{i} \widetilde{\alpha}_{i} \widetilde{r}_{i} \quad \text{if} \quad \widetilde{\lambda}_{i} < 0$$
 (33a)

or

$$\mathbf{w}_{\mathbf{j}}^{\mathbf{n}+1} = \mathbf{w}_{\mathbf{j}}^{\mathbf{n}} - \frac{\Delta t}{J_{\mathbf{j}-\mathbf{k}}\Delta \xi} \widetilde{\lambda}_{\mathbf{i}} \widetilde{\alpha}_{\mathbf{i}} \widetilde{\Sigma}_{\mathbf{i}} \quad \text{if} \quad \widetilde{\lambda}_{\mathbf{i}} > 0 . \tag{33b}$$

Extensions of this first order algorithm to second order can be made [5] and to non-uniform grids [6, 7].

#### 3.4 Grid generation and grid Jacobian

The purpose of this paper is to present an efficient Riemann solver for use with non-Cartesian body-fitted coordinates. The mapping from physical (x, y, z) space to computational  $(\xi, \eta, \zeta)$  space can be given analytically, or constructed numerically [8]. In the case where the mapping  $x = x(\xi, \eta, \zeta)$ ,  $y = y(\xi, \eta, \zeta)$ ,  $z = z(\xi, \eta, \zeta)$  is known analytically we can approximate  $J_{j-\frac{1}{2}}$  in equation (30) as

$$J_{1-\frac{1}{2}} = J(\xi_{1-\frac{1}{2}}, \eta_0, \zeta_0) ; \qquad (34)$$

alternatively,  $J_{j-1/2}$  can be approximated using central; differences. In addition, we will need a suitable approximation  $x_{\eta}^{j-1/2}$  for  $x_{\eta}$  at  $(\xi_{j-1/2}, \eta_0, \zeta_0)$  and in the analytic case we take

$$x_{\eta}^{j-1/2} = x_{\eta}(\xi_{j-1/2}, \eta_0, \zeta_0)$$
 (35)

as in equation (34); otherwise we set  $x_{\eta}^{j-k}$  to be the arithmetic mean of central difference approximations to  $x_{\eta}$  at  $(\xi_{j-1}, \eta_0, \zeta_0)$  and  $(\xi_j, \eta_0, \zeta_0)$ . Similar approximations hold for  $y_{\eta}, x_{\xi}, x_{\zeta}$  etc., and these give rise to an obvious approximation  $(J_{\xi}^{x})_{j-k}$  for  $J_{\xi}^{x}$ , etc.

# 3.5 Construction of $\widetilde{A}_{j-1/2}$

Consider a  $\xi$  coordinate line given by  $\eta = \eta_0$ ,  $\zeta = \zeta_0$ , constants, and denote points  $\xi_{j-1}$ ,  $\xi_j$  on this line by  $\xi_L$ ,  $\xi_R$ , respectively. In addition, we denote  $w_{j-1}^n = w_L$ ,  $w_j^n = w_R$ , and assume that  $X = (J_\xi^x)_{j-1/2}$ ,  $Y = (J_\xi^y)_{j-1/2}$  and  $Z = (J_\xi^z)_{j-1/2}$  denote approximations to  $J_\xi^x$ ,  $J_\xi^y$  and  $J_\xi^z$ , respectively, that are constant in the interval  $(\xi_L, \xi_R)$ . Our aim is to construct a matrix  $\widetilde{A}_{j-1/2} = \widetilde{A}(w_L, w_R)$  satisfying properties (i) and (ii) of §3.2. Equivalently, we could find average eigenvalues  $\widetilde{\lambda}_i$  and average eigenvectors  $\widetilde{\tau}_i$  of the eigenvalues and eigenvectors of the Jacobian matrix A at  $\xi_L$ ,  $\xi_R$  given by equations (20a)-(22) such that

$$\Delta_{\mathbf{w}} = \sum_{i=1}^{5} \widetilde{\alpha}_{i} \widetilde{z}_{i}$$
 (36a-e)

and

$$\Delta_{\widetilde{\Sigma}}^{F} = \sum_{i=1}^{5} \widetilde{\lambda}_{i} \widetilde{\alpha}_{i} \widetilde{\widetilde{\Sigma}}_{i}$$
 (37a-e)

for some wavestrengths  $\stackrel{\sim}{lpha}_{\mathbf{i}}$  , where

$$\Delta(\cdot) = (\cdot)_{R} - (\cdot)_{L} . \tag{38}$$

This yields the following approximate Jacobian matrix

$$\widetilde{A}_{j-\frac{1}{2}} = \widetilde{M}_{j-\frac{1}{2}}\widetilde{D}_{j-\frac{1}{2}}\widetilde{M}_{j-\frac{1}{2}}^{-1}$$
(39)

#### 3.6 Wavespeeds for nearby states

Consider two (constant) adjacent states  $w_L$ ,  $w_R$  (left and right) close to an average state w, at points L and R on a  $\xi$ -coordinate line. In particular, the variable  $\gamma$  given by equation (25) is piecewise constant. Now in view of the sound speed a for ideal gases ( $\gamma \equiv \text{constant}$ ) given by equation (28) and the eigenvectors  $e_{1,2}$  given by equations (27a-b), we assume that we have approximate eigenvectors

$$\mathcal{E}_{1,2} = \left[1, u \pm \frac{aX}{d}, v \pm \frac{aY}{d}, w \pm \frac{aZ}{d}, \frac{a^2}{\gamma - 1} + \frac{1}{2}(u^2 + v^2 + w^2) \pm \frac{aU}{d}\right]^T$$
(40a-b)

where X, Y and Z are as described in §3.5,

$$d = \sqrt{X^2 + Y^2 + Z^2} \tag{41}$$

and

$$U = Xu + Yv + Zw (42)$$

corresponding to the average state w . (N.B. The quantity  $\gamma$  in

equations (40a-b) represents an average value close to  $\gamma_L$  and  $\gamma_R$ .) In addition, because  $i-\frac{\rho p}{p_i}=0$  for an ideal gas, we split  $e_3$  into two vectors as

$$r_3' = \left[1, u, v, w, \frac{1}{2}(u^2 + v^2 + w^2)\right]^T$$
 (43)

and

$$r_{6}' = (0, 0, 0, 0, \beta)^{T},$$
 (44)

where  $\beta$  represents an average value in the cell  $(\xi_L, \xi_R)$  of  $i-\rho p_\rho/p_i$  . Finally, we approximate  $e_{4.5}$  as

$$r_4 = (0, -Y, X, 0, Xv - Yu)^T$$
 (45)

and

$$\mathbf{r}_{5} = (0, -Z, 0, X, Xw - Zu)^{T}$$
 (46)

We seek coefficients  $\alpha_1$ ,  $\alpha_2$ ,  $\alpha_3$ ,  $\alpha_4$ ,  $\alpha_5$  such that

$$\Delta_{w} = \alpha_{1}r_{1} + \alpha_{2}r_{2} + \alpha_{3}r_{3} + \alpha_{4}r_{4} + \alpha_{5}r_{5} + r_{6}$$
 (47)

to within  $O(\Delta^2)$ . (N.B. The vector  $\mathbf{r}_6$  is considered separately since it vanishes for an ideal gas. Also, we do not introduce another coefficient  $\alpha_6$  since  $\mathbf{r}_6$  has only one non-zero component and is therefore not required.) After some manipulation (see Appendix A) we find that equation (47) yields the following expressions for  $\alpha_1$  and  $\beta$ 

$$\alpha_{1,2} = \frac{1}{2a^2} \left[ \Delta p \pm \frac{\rho a \Delta U}{d} \right]^T \tag{48a-b}$$

$$\alpha_3 = \Delta \rho - \frac{\Delta p}{a^2} \tag{48c}$$

$$\alpha_4 = \frac{\rho \Delta v}{X} - \frac{\rho Y \Delta U}{d^2 X} \tag{48d}$$

$$\alpha_{5} = \frac{\rho \Delta w}{X} - \frac{\rho Z \Delta U}{d^{2} X}$$
 (48e)

and

$$\beta = -\frac{\rho i \Delta \gamma}{\gamma - 1} \qquad (49)$$

where we have made the assumption that to within  $O(\Delta^2)$ 

$$\Delta(\rho N) = N\Delta\rho + \rho\Delta N$$
,  $N = u, v \text{ or } w$  (50a-c)

$$\Delta(\rho N^2) = N^2 \Delta \rho + 2\rho N \Delta N$$
,  $N = u, v \text{ or } w$  (51a-c)

and

$$\Delta \left[ \frac{p}{\gamma - 1} \right] = \frac{\Delta p}{\gamma - 1} - \frac{p}{(\gamma - 1)^2} \Delta \gamma \qquad (52)$$

The results above imply an approximation to the eigenvector  $e_3$  given by  $r_3 = r_3' + r_6'/\alpha_3$  i.e.

$$r_3 = \left[1, u, v, w, \frac{1}{2}(u^2+v^2+w^2) - \frac{\rho i \Delta \gamma}{(\gamma-1)\left[\Delta \rho - \frac{\Delta p}{a^2}\right]}\right]^T$$
, (53)

and hence an approximation to  $i - \rho p_{\rho}/p_{i}$  . With the expressions

given above it is possible to show that

$$\Delta \mathbf{\tilde{F}} = \sum_{i=1}^{5} \lambda_{i} \alpha_{i} \mathbf{\tilde{r}}_{i}$$
 (54)

to within  $O(\Delta^2)$  . We now return to the general case.

# 3.7 Decomposition for general $\underset{L}{w}_L$ , $\underset{R}{w}_R$

Consider two states  $\underset{L}{w}_L$ ,  $\underset{R}{w}_R$  not necessarily close such that equations (36a)-(37e) are satisfied exactly, where

$$\tilde{\lambda}_{i} = \tilde{U} \pm \tilde{ad}, \tilde{U}, \tilde{U}, \tilde{U}$$
 (55a-e)

$$\vec{r}_{1,2} = \begin{bmatrix} 1, \vec{u} \pm \frac{\vec{a}X}{d}, \vec{v} \pm \frac{\vec{a}Y}{d}, \vec{w} \pm \frac{\vec{a}Z}{d}, \frac{\vec{a}^2}{\vec{\gamma}-1} + \frac{\vec{\omega}}{\vec{u}^2} \end{bmatrix}$$

$$+ \frac{\tilde{v}^2}{4\tilde{v}^2} + \frac{\tilde{w}^2}{4\tilde{v}^2} \pm \frac{\tilde{a}\tilde{U}}{\tilde{d}} \right]^{T}$$
 (56a-b)

$$\tilde{r}_{3} = \begin{bmatrix} 1, & \tilde{u}, & \tilde{v}, & \tilde{w}, & \frac{\tilde{u}^{2} + \tilde{v}^{2} + \tilde{w}^{2}}{\tilde{v}^{2} + \tilde{u}^{2}} \end{bmatrix} - \frac{\tilde{\rho} i \Delta \tau}{\tilde{(\tau - 1)} \left[ \Delta \rho - \frac{\Delta p}{\tilde{a}^{2}} \right]} \end{bmatrix}^{T}$$
 (56c)

$$\vec{r}_{4} = (0, -Y, X, 0, Xv - Yu)$$
 (56d)

$$r_{5} = (0, -Z, 0, X, X_{w} - Z_{u})^{T}$$
 (56e)

$$\overset{\sim}{\alpha}_{1,2} = \frac{1}{2\overset{\sim}{a^2}} \left[ \Delta p \pm \frac{\overset{\sim}{\rho a \Delta U}}{d} \right]$$
 (57a-b)

$$\overset{\sim}{\alpha}_{3} = \Delta \rho - \frac{\Delta p}{\overset{\sim}{\alpha}^{2}}$$
(57d)

$$\widetilde{\alpha}_{4} = \frac{\widetilde{\rho}\Delta v}{X} - \frac{\widetilde{\rho}Y\Delta U}{d^{2}X}$$
 (57d)

$$\widetilde{\alpha}_{5} = \frac{\widetilde{\rho}\Delta w}{X} - \frac{\widetilde{\rho}Z\Delta U}{d^{2}X}$$
 (57e)

$$\widetilde{U} = Xu + Yv + Zw$$
 (58)

$$\gamma = \frac{p(\rho, i)}{\rho i} + 1 \tag{59}$$

and

$$\Delta U = \Delta (Xu + Yv + Zw) = X\Delta u + Y\Delta v + Z\Delta w . \tag{60}$$

(N.B. X, Y and Z are constant in  $(\xi_L, \xi_R)$ .) Thus, we have to determine averages  $\rho$ , u, v, w, a, i, and  $\gamma$  such that equations (36a)-(37e) are satisfied subject to equations (55a)-(60). The solution to this problem can be determined (see Appendix B) and gives the following averages

$$\widetilde{N} = \frac{\sqrt{\rho_L} N_L + \sqrt{\rho_R} N_R}{\sqrt{\rho_L} + \sqrt{\rho_R}}, \quad N = u, v, w, i, \gamma \text{ or } H \quad (61a-f)$$

$$\stackrel{\sim}{\rho} = \sqrt{\rho_L \rho_R} \tag{62}$$

$$\stackrel{\sim}{a^2} = \stackrel{\sim}{(\gamma - 1)} \left[ \stackrel{\sim}{H} - \frac{\omega}{(u^2 + v^2 + w^2)} \right] , \qquad (63)$$

where

$$H = \frac{p}{\rho} + i + \frac{1}{2}(u^{2} + v^{2} + w^{2})$$

$$= \frac{\gamma p}{\rho(\gamma - 1)} + \frac{1}{2}(u^{2} + v^{2} + w^{2})$$
(64)

is the enthalpy.

Similar results hold for updating in the  $\eta$  and  $\zeta$  directions.

In the next section we give the numerical results for a test problem using the algorithm of this section.

#### 4. NUMERICAL RESULTS

In this section we give the numerical results for a standard test problem in two-dimensional gas dynamics using the Riemann solver described in §3.

The problem is that of uniform flow of 'real air' past a circular cylinder. The equation of state used can be written as

$$p = \left[ \gamma(\rho, i) - 1 \right] \rho i$$

where the form of  $\gamma(\rho, i)$  is determined via curve fits to experimental data [3]. The radius of the cylinder is 0.5 and the initial conditions are  $\rho=1.4$ , u=8.0, v=w=0.0, p=1.0 which corresponds to Mach 8 flow. An 0-type computational mesh is used and thus the grid transformation is from (x, y, z) physical space to  $(\xi, \eta, \zeta) \equiv (R, \phi, z)$  computational space, where  $R, \phi, z$  are standard cylindrical polar coordinates. The region of computation considered is  $(R, \phi) \in [0.5, 20] \times [0, 2\pi]$  and we apply periodic conditions along  $\phi=0$ .

The grid spacing in the  $\phi$ -direction is uniform with 128 grid lines given by  $\phi_j=(j-1/2)\frac{\pi}{128}$ ,  $j=1,\ldots,128$ . In the R-direction the grid spacing is geometric with 33 grid lines given by  $R_1=0.5+1/2$ ,  $R_j=R_{j-1}+1/2$ ,  $R_j=1/2$ ,  $R_j=1/2$ , where  $R_j=1/2$  and  $R_j=1/2$ . Figure 1 displays the isomach contours for Mach 8

flow after 1000 time steps in the same region. The complete grid is shown in figure 2 and that part of the grid in  $0.5 \le R \le 5.0$  is shown in figure 3. Figure 4 shows the complete grid with the portion shown in figure 3 removed.

In all cases the shock has been captured over at most three cells.

#### 5. CONCLUSIONS

We have presented an efficient Riemann solver for three-dimensional compressible flows using body-fitted coordinates. The scheme applies to a general convex equation of state and by using a local parameterisation of the equation of state, only one function call is required per one-dimensional computational cell. The numerical results achieved show that the shock has been captured over two or three cells.

#### ACKNOWLEDGEMENTS

I acknowledge the financial support of A.W.R.E., Aldermaston.

#### REFERENCES

- 1. P.L. ROE 'Approximate Riemann Solvers, Parameter Vectors and Difference Schemes', <u>J. Comput. Phys.</u> 49, 357 (1983).
- 2. P. GLAISTER, 'An Approximate Linearised Riemann Solver for the Euler equations in One Dimension for Real Gases', J. Comput. Phys. 74, 2, 382 (1988).
- 4. S.K. GODUNOV, 'A Difference Method for the Numerical Computation of Continuous Solutions of Hydrodynamic Equations', Mat. Sbornik, 47, no. 3, 271, (1959).
- 5. P.K. SWEBY, 'High Resolution Schemes using Flux Limiters for Hyperbolic Conservation Laws', <u>SIAM J. Numer. Anal.</u> 21, 995, (1984).
- 6. P. GLAISTER 'Shock Capturing on Irregular Grids', Numerical Analysis Report 4-86, University of Reading. (1986).
- 7. J. PIKE 'Grid Adaptive Algorithms for the Solution of the Euler equations on Irregular Grids', <u>J. Comput. Phys.</u> 71, 194, (1987).
- 8. J.F. THOMPSON, Z.U.A. WARSI and C.W. MASTIN,
  'Numerical Grid Generation Foundations and
  Applications', North-Holland, (1985).

#### DESCRIPTIVE LEGENDS

Figure 1 Isomach contours for Mach 8 flow in  $0.5 \le R \le 5.0$ 

Figure 2 The complete grid

Figure 3 The grid in  $0.5 \le R \le 5.0$ 

Figure 4 The complete grid with the portion shown in figure 4 removed

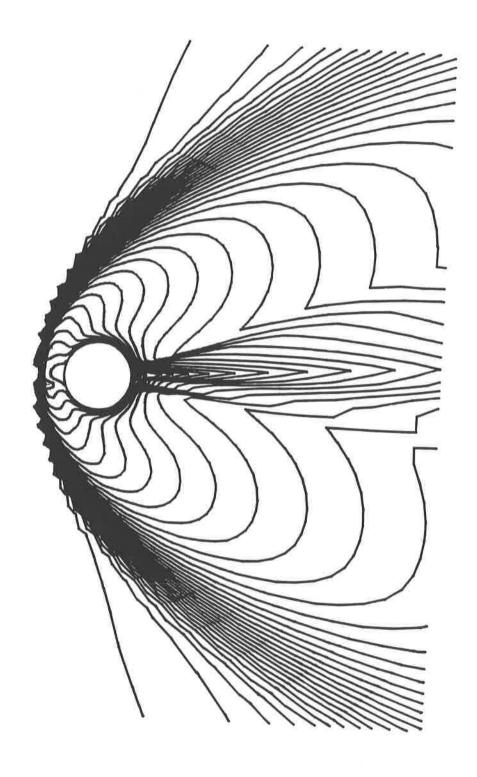


Figure 1

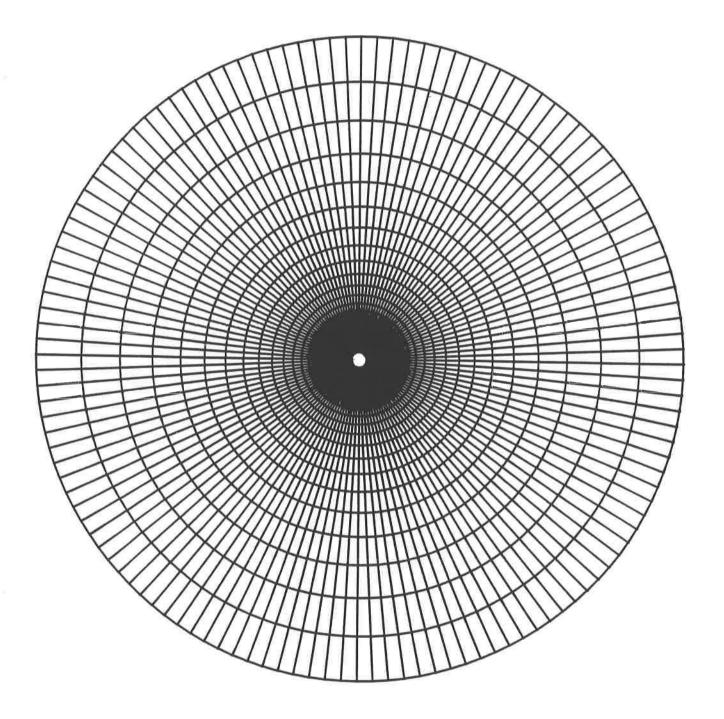
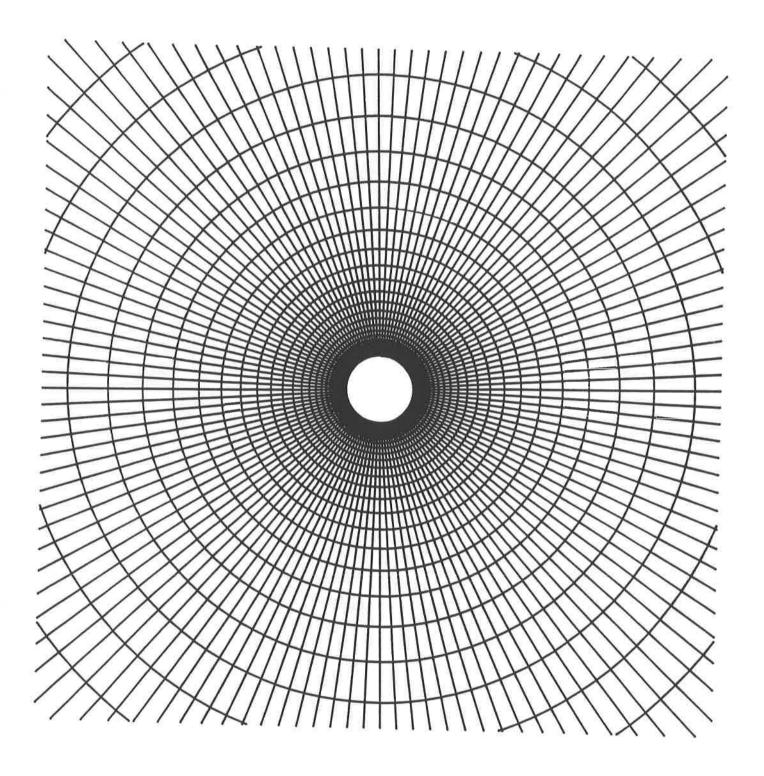


Figure 2



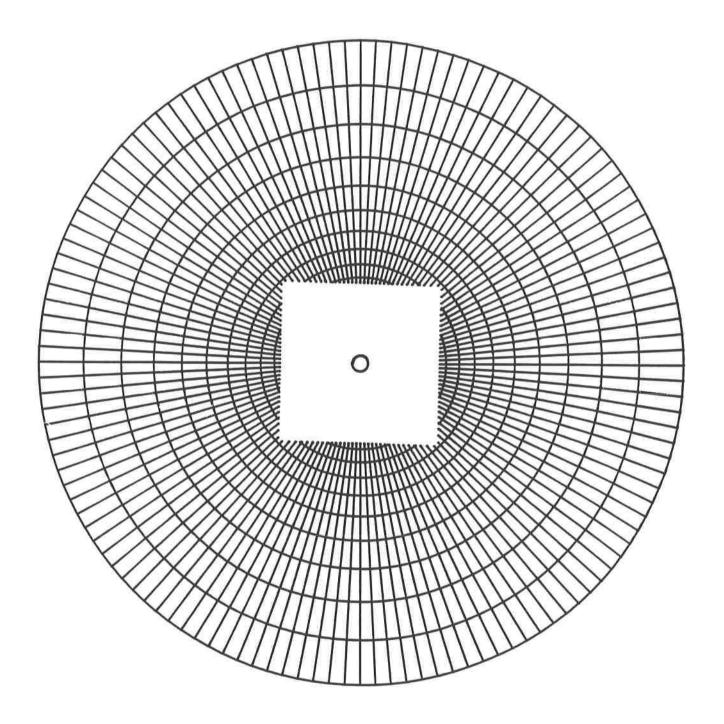


Figure 4

#### APPENDIX A

In this appendix we derive the expressions given in equations (48a)-(49) for the wavestrengths  $\alpha_i$  and the term  $\beta$  so that equations (47) are satisfied to within  $O(\Delta^2)$ . Writing equations (47) out in full we have

$$\Delta \rho = \alpha_1 + \alpha_2 + \alpha_3 \tag{A1}$$

$$\Delta(\rho \mathbf{u}) = \alpha_1 \left[ \mathbf{u} + \frac{\mathbf{a} \mathbf{X}}{\mathbf{d}} \right] + \alpha_2 \left[ \mathbf{u} - \frac{\mathbf{a} \mathbf{X}}{\mathbf{d}} \right] + \alpha_3 \mathbf{u} - \alpha_4 \mathbf{Y} - \alpha_5 \mathbf{Z}$$
 (A2)

$$\Delta(\rho v) = \alpha_1 \left[ v + \frac{aY}{d} \right] + \alpha_2 \left[ v - \frac{aY}{d} \right] + \alpha_3 v + \alpha_4 X \tag{A3}$$

$$\Delta(\rho w) = \alpha_1 \left[ w + \frac{aZ}{d} \right] + \alpha_2 \left[ w - \frac{aZ}{d} \right] + \alpha_3 w + \alpha_5 X \tag{A4}$$

$$\Delta e = \Delta(\rho i) + \Delta \left[ \rho \frac{u^2}{2} \right] + \Delta \left[ \rho \frac{v^2}{2} \right] + \Delta \left[ \rho \frac{w^2}{2} \right]$$

$$= \alpha_1 \left[ \frac{a^2}{\gamma - 1} + \frac{1}{2} (u^2 + v^2 + w^2) + \frac{aU}{d} \right]$$

$$+ \alpha_2 \left[ \frac{a^2}{\gamma - 1} + \frac{1}{2} (u^2 + v^2 + w^2) - \frac{aU}{d} \right]$$

$$+ \alpha_3 \frac{1}{2} (u^2 + v^2 + w^2) + \alpha_4 (Xv - Yu) + \alpha_5 (Xw - Zu) + \beta$$
(A5)

where

$$d = \sqrt{X^2 + Y^2 + Z^2}$$
 (A6)

and

$$U = Xu + Yv + Zw . (A7)$$

From equations (A1)-(A4) we have

$$\Delta(\rho u) - u\Delta \rho = \frac{aX}{d} (\alpha_1 - \alpha_2) - \alpha_4 Y - \alpha_5 Z \tag{A8}$$

$$\Delta(\rho v) - v\Delta \rho = \frac{aY}{d} (\alpha_1 - \alpha_2) + \alpha_4 X$$
 (A9)

$$\Delta(\rho w) - w\Delta \rho = \frac{aZ}{d} (\alpha_1 - \alpha_2) + \alpha_5 X \qquad (A10)$$

and combining equations (A8), (A9) and (A10) we obtain

$$\Delta(\rho U) - U\Delta \rho = ad(\alpha_1 - \alpha_2)$$
 (A11)

since U = Xu + Yv + Zw and X, Y, Z are constant.

If we assume

$$\Delta(\rho N) = \rho \Delta N + N \Delta \rho$$
,  $N = u, v, w$  (A12)

to within  $O(\Delta^2)$  , then equation (A11) yields

$$ad(\alpha_1 - \alpha_2) = \rho \Delta U \tag{A13}$$

and hence from equations (A9), (A10), (A11) and (A12) we find that

$$\alpha_4 = \frac{\rho \Delta v}{X} - \frac{\rho Y \Delta U}{d^2 X}$$
 (A14)

and

$$\alpha_{5} = \frac{\rho \Delta w}{X} - \frac{\rho Z \Delta U}{d^{2} X} . \tag{A15}$$

Combining equations (A1), (A8), (A9) and (A10) equation (A5) yields

$$\Delta(\rho i) + \Delta \left[\frac{\rho u^{2}}{2}\right] + \Delta \left[\frac{\rho v^{2}}{2}\right] + \Delta \left[\frac{\rho w^{2}}{2}\right] 
= \frac{a^{2}}{\gamma - 1}(\alpha_{1} + \alpha_{2}) + \frac{1}{2}(u^{2} + v^{2} + w^{2})\Delta\rho + \frac{aU}{d}(\alpha_{1} - \alpha_{2}) 
+ u \left[\Delta(\rho u) - u\Delta\rho\right] + v \left[\Delta(\rho v) - v\Delta\rho\right] + w \left[\Delta(\rho w) - w\Delta\rho\right] 
- \frac{a}{d}U(\alpha_{1} - \alpha_{2}) + \beta .$$
(A16)

Now, in view of equation (A12) together with the relations

$$\Delta(\rho N^2) = N^2 \Delta \rho + 2\rho N \Delta N , \qquad N = u, v, w \qquad (A17)$$

to within  $O(\Delta^2)$  , then equation (A16) gives

$$\Delta(\rho i) = \frac{a^2}{\gamma - 1} (\alpha_1 + \alpha_2) + \beta . \tag{A18}$$

However, we also know that  $\rho i = \frac{p}{\gamma - 1}$ , therefore

$$\Delta(\rho i) = \Delta\left[\frac{p}{\gamma-1}\right] = \frac{\Delta p}{\gamma-1} - \frac{p}{(\gamma-1)^2} \Delta\gamma$$

$$= \frac{\Delta p}{\gamma-1} - \frac{\rho i}{\gamma-1} \Delta\gamma \qquad (A19)$$

to within  $O(\Delta^2)$  , and thus using equation (A1), equation (A18) gives

$$\alpha_3 - \left[\Delta \rho - \frac{\Delta p}{a^2}\right] = \frac{(\gamma - 1)}{a^2} \left[\beta + \frac{\rho i \Delta \gamma}{\gamma - 1}\right].$$
 (A20)

In the ideal case  $\gamma \equiv \text{constant}$ , so that  $\Delta \gamma = 0$  and  $i - \rho p_{\rho}/p_{i} = 0$ , i.e.  $\beta = 0$  for consistency and thus  $\alpha_{3} = \Delta \rho - \frac{\Delta p}{a^{2}}$ . In the general case we would like  $\alpha_{3} \to \Delta \rho - \frac{\Delta p}{a^{2}}$  as  $\Delta \gamma \to 0$ , and thus we set

$$\beta = -\frac{\rho i \Delta \gamma}{\gamma - 1} \tag{A21}$$

so that

$$\alpha_{3} = \Delta \rho - \frac{\Delta p}{a^{2}} . \tag{A22}$$

Finally, equations (A1), (A13) and (A22) yield

$$\alpha_{1,2} = \frac{1}{2a^2} \left[ \Delta p \pm \frac{\rho a \Delta U}{d} \right] \qquad (A23)$$

#### APPENDIX B

In this appendix we derive the averages that make equations (36a)-(37e) satisfied.

Writing equations (36a)-(37e) out in full we have

$$\Delta \rho = \alpha_1^2 + \alpha_2^2 + \alpha_3^2 \tag{B1}$$

$$\Delta(\rho \mathbf{u}) = \alpha_1 \left[ \mathbf{u} + \frac{\mathbf{a} \mathbf{X}}{\mathbf{d}} \right] + \alpha_2 \left[ \mathbf{u} - \frac{\mathbf{a} \mathbf{X}}{\mathbf{d}} \right] + \alpha_3 \mathbf{u} - \alpha_4 \mathbf{Y} - \alpha_5 \mathbf{Z}$$
 (B2)

$$\Delta(\rho \mathbf{v}) = \overset{\sim}{\alpha_1} \left[ \overset{\sim}{\mathbf{v}} + \frac{\overset{\sim}{\mathbf{a} \mathbf{Y}}}{\mathbf{d}} \right] + \overset{\sim}{\alpha_2} \left[ \overset{\sim}{\mathbf{v}} - \frac{\overset{\sim}{\mathbf{a} \mathbf{Y}}}{\mathbf{d}} \right] + \overset{\sim}{\alpha_3} \overset{\sim}{\mathbf{v}} + \overset{\sim}{\alpha_4} \mathbf{X}$$
 (B3)

$$\Delta(\rho w) = \alpha_1 \left[ \widetilde{w} + \frac{\widetilde{a}Z}{d} \right] + \alpha_2 \left[ \widetilde{w} - \frac{\widetilde{a}Z}{d} \right] + \alpha_3 \widetilde{w} + \alpha_5 X$$
 (B4)

$$\Delta e = \Delta \left[ \rho \mathbf{i} + \frac{\rho \mathbf{q}^2}{2} \right] = \alpha_1 \left[ \frac{\alpha^2}{\gamma_{-1}} + \frac{\alpha^2}{2} + \frac{\widetilde{aU}}{d} \right]$$

$$+ \quad \overset{\sim}{\alpha}_2 \left[ \frac{\overset{\sim}{\alpha^2}}{\overset{\sim}{\gamma-1}} + \overset{\sim}{\cancel{M}} \overset{\sim}{q^2} - \frac{\overset{\sim}{\alpha U}}{\overset{\sim}{d}} \right] + \overset{\sim}{\alpha}_3 \overset{\sim}{\cancel{M}} \overset{\sim}{q^2} - \frac{\overset{\sim}{\rho i \Delta \gamma}}{\overset{\sim}{\gamma-1}}$$

+ 
$$\alpha_4^{\sim}(Xv-Yu)$$
 +  $\alpha_5^{\sim}(Xw-Zu)$  (B5)

$$\Delta(\rho U) = \alpha_1 (\ddot{U} + ad) + \alpha_2 (\ddot{U} - ad) + \alpha_3 \ddot{U}$$
 (B6)

$$\Delta(Xp + \rho uU) = \alpha_1 (\widetilde{U} + \widetilde{ad}) \left[\widetilde{u} + \frac{\widetilde{aX}}{d}\right] + \alpha_2 (\widetilde{U} - \widetilde{ad}) \left[\widetilde{u} - \frac{\widetilde{aX}}{d}\right] + \alpha_3 \widetilde{U}\widetilde{u} - \alpha_4 \widetilde{U}Y - \alpha_5 \widetilde{U}Z$$
(B7)

$$\Delta(\mathrm{Yp}+\rho\mathrm{vU}) = \overset{\sim}{\alpha_1}(\overset{\sim}{\mathrm{U}}+\overset{\sim}{\mathrm{ad}})\left[\overset{\sim}{\mathrm{v}} + \overset{\simeq}{\overset{\simeq}{\mathrm{aY}}}\right] + \overset{\sim}{\alpha_2}(\overset{\sim}{\mathrm{U}}-\overset{\sim}{\mathrm{ad}})\left[\overset{\sim}{\mathrm{v}} - \overset{\simeq}{\overset{\simeq}{\mathrm{aY}}}\right] + \overset{\sim}{\alpha_3}\overset{\sim}{\mathrm{Uv}} + \overset{\sim}{\alpha_4}\overset{\sim}{\mathrm{UX}}$$
(B8)

$$\begin{split} \Delta(\mathsf{Zp+\rho wU}) &= \overset{\sim}{\alpha_1} (\overset{\sim}{\mathsf{U}+ad}) \left[\overset{\sim}{\mathsf{w}} + \frac{\overset{\sim}{\mathsf{a}Z}}{\mathsf{d}}\right] + \overset{\sim}{\alpha_2} (\overset{\sim}{\mathsf{U}-ad}) \left[\overset{\sim}{\mathsf{w}} - \frac{\overset{\sim}{\mathsf{a}Z}}{\mathsf{d}}\right] \\ &+ \overset{\sim}{\alpha_3} \overset{\sim}{\mathsf{Uw}} + \overset{\sim}{\alpha_5} \overset{\sim}{\mathsf{UX}} \end{split} \tag{B9}$$

$$\Delta\left[\mathsf{U}(\mathsf{e+p})\right] &= \Delta\left(\mathsf{U}\rho\,\mathsf{i}+\mathsf{Up}\right) + \Delta\left[\rho\frac{\mathsf{Uq}^2}{2}\right] \\ &= \overset{\sim}{\alpha_1} (\overset{\sim}{\mathsf{U}+ad}) \left[\overset{\sim}{\overset{\sim}{\mathsf{w}^2}} + \overset{\sim}{\mathsf{W}}\overset{\sim}{\mathsf{q}^2} + \frac{\overset{\sim}{\mathsf{a}U}}{\mathsf{d}}\right] \\ &+ \overset{\sim}{\alpha_2} (\overset{\sim}{\mathsf{U}-ad}) \left[\overset{\sim}{\overset{\sim}{\mathsf{w}^2}} + \overset{\sim}{\mathsf{W}}\overset{\sim}{\mathsf{q}^2} - \frac{\overset{\sim}{\mathsf{a}U}}{\mathsf{d}}\right] \\ &+ \overset{\sim}{\alpha_3} \overset{\sim}{\mathsf{U}}\overset{\sim}{\mathsf{W}}\overset{\sim}{\mathsf{q}^2} - \frac{\overset{\sim}{\mathsf{U}}\overset{\sim}{\mathsf{p}^1}\Delta\gamma}{\overset{\sim}{\gamma-1}} \\ &+ \overset{\sim}{\alpha_4} \overset{\sim}{\mathsf{U}} (\overset{\sim}{\mathsf{X}^2}-\overset{\sim}{\mathsf{V}}) + \overset{\sim}{\alpha_5} \overset{\sim}{\mathsf{U}} (\overset{\sim}{\mathsf{X}^2}-\overset{\sim}{\mathsf{U}}) \end{aligned} \tag{B10}$$

where

$$U = Xu + Yv + Zw$$
 (B11)

$$d = \sqrt{X^2 + Y^2 + Z^2}$$
 (B12)

$$q^2 = u^2 + v^2 + w^2$$
 (B13)

and for convenience we have written

$$\vec{q}^2 = \vec{u}^2 + \vec{v}^2 + \vec{w}^2$$
 (B14)

In addition, X, Y and Z are constant and  $\alpha_i$ ,  $i=1,\ldots,5$  are given by

$$\tilde{\alpha}_{1,2} = \frac{1}{2\tilde{a}^2} \left[ \Delta p \pm \frac{\tilde{\rho} a \Delta U}{d} \right]$$
 (B15a-b)

$$\overset{\sim}{\alpha_3} = \Delta \rho - \frac{\Delta p}{\tilde{\alpha}^2}$$
 (B16)

$$\tilde{\alpha}_{4} = \frac{\tilde{\rho}\Delta v}{X} - \frac{\tilde{\rho}Y\Delta U}{d^{2}X}$$
(B17)

and

$$\overset{\sim}{\alpha}_{5} = \frac{\overset{\sim}{\rho \Delta w}}{X} - \frac{\overset{\sim}{\rho Z \Delta U}}{d^{2} X} . \tag{B18}$$

Firstly, equation (B1) is satisfied by any average, and noting that

$$\Delta U = \Delta (Xu+Yv+Zw) = X\Delta u + Y\Delta v + Z\Delta w , \qquad (B19)$$

equations (B1)-(B4) yield

$$\Delta(\rho \mathbf{u}) = \rho \Delta \mathbf{u} + \mathbf{u} \Delta \rho$$
 (B20)

$$\Delta(\rho \mathbf{v}) = \stackrel{\sim}{\rho \Delta \mathbf{v}} + \stackrel{\sim}{\mathbf{v} \Delta \rho} \tag{B21}$$

$$\Delta(\rho w) = \stackrel{\sim}{\rho} \Delta w + \stackrel{\sim}{w} \Delta \rho$$
 (B22)

whilst equation (B6) gives

$$\Delta(\rho U) = \rho \Delta U + U \Delta \rho \qquad (B23)$$

From equations (B7), (B8) and (B9) we obtain, respectively,

$$\Delta(\rho u U) = u U \Delta \rho + \rho u \Delta U + \rho U \Delta u$$
 (B24)

$$\Delta(\rho v U) = v U \Delta \rho + \rho v \Delta U + \rho U \Delta v$$
 (B25)

$$\Delta(\rho wU) = WU\Delta\rho + \rho W\Delta U + \rho U\Delta w$$
 (B26)

which combine to give

$$\Delta(\rho U^2) = \widetilde{U}^2 \Delta \rho + 2\rho U \Delta U . \qquad (B27)$$

Substituting for  $\stackrel{\sim}{\rho}$  from equation (B23) into equation (B27) yields the following quadratic equation for  $\stackrel{\sim}{U}$ 

$$\widetilde{U}^{2}\Delta\rho - 2\widetilde{U}\Delta(\rho U) + \Delta(\rho U^{2}) = 0 \qquad (B28)$$

Only one solution of equation (B28) is productive, namely

$$\widetilde{U} = \frac{\Delta(\rho U) - \sqrt{\left[\Delta(\rho U)\right]^2 - \Delta\rho\Delta(\rho U^2)}}{\Delta\rho}$$
(B29)

i.e.

$$\widetilde{U} = \frac{\sqrt{\rho_L} U_L + \sqrt{\rho_R} U_R}{\sqrt{\rho_L} + \sqrt{\rho_R}} , \qquad (B30)$$

and equation (B23) now gives

$$\stackrel{\sim}{\rho} = \frac{\Delta(\rho U) - \stackrel{\sim}{U}\Delta\rho}{\Delta U} = \sqrt{\rho_L \rho_R} . \tag{B31}$$

Equations (B20), (B21) and (B22) give

$$\tilde{\mathbf{u}} = \frac{\Delta(\rho \mathbf{u}) - \rho \Delta \mathbf{u}}{\Delta \rho} = \frac{\sqrt{\rho_{L}} \mathbf{u}_{L} + \sqrt{\rho_{R}} \mathbf{u}_{R}}{\sqrt{\rho_{L}} + \sqrt{\rho_{R}}} , \qquad (B32)$$

$$\tilde{v} = \frac{\Delta(\rho v) - \rho \Delta v}{\Delta \rho} = \frac{\sqrt{\rho_L} v_L + \sqrt{\rho_R} v_R}{\sqrt{\rho_L} + \sqrt{\rho_R}}$$
(B33)

and

$$\widetilde{\mathbf{w}} = \frac{\Delta(\rho \mathbf{w}) - \rho \Delta \mathbf{w}}{\Delta \rho} = \frac{\sqrt{\rho_{\mathbf{L}}} \mathbf{w}_{\mathbf{L}} + \sqrt{\rho_{\mathbf{R}}} \mathbf{w}_{\mathbf{R}}}{\sqrt{\rho_{\mathbf{L}}} + \sqrt{\rho_{\mathbf{R}}}} . \tag{B34}$$

We note that equations (B11), (B30) and (B32)-(B34) imply that

$$\widetilde{U} = X\widetilde{u} + Y\widetilde{v} + Z\widetilde{w} . \tag{B35}$$

We have now determined  $\overset{\sim}{\rho}$ ,  $\overset{\sim}{u}$ ,  $\overset{\sim}{v}$  and  $\overset{\sim}{w}$ , so that we are left with equations (B5) and (B10) to determine  $\overset{\sim}{a}$ ,  $\overset{\sim}{\gamma}$  and  $\overset{\sim}{i}$ . We begin by employing equations (B15a)-(B18) so that equation (B10) becomes

$$\Delta(\rho i) + \Delta\left[\frac{\rho q^{2}}{2}\right] = \frac{\Delta p}{\widetilde{\gamma}-1} + \frac{2}{2}\widetilde{q}^{2}\Delta\rho - \frac{\widetilde{\rho} i\Delta\gamma}{\widetilde{\gamma}-1} + \widetilde{\rho}(\widetilde{u}\Delta u + v\Delta v + \widetilde{w}\Delta w) \qquad (B36)$$

Now, it is a simple matter to verify that

$$\Delta \left[ \frac{\rho q^2}{2} \right] = \frac{\kappa q^2 \Delta \rho}{2} + \frac{\kappa q^2 \Delta \rho}{\rho (u \Delta u + v \Delta v + w \Delta w)}$$
 (B37)

so that equation (B36) gives, after rearrangement,

$$(\gamma-1)\Delta(\rho i) + \rho i\Delta\gamma - \Delta p = 0 .$$
 (B38)

If we define averages  $\hat{\gamma}$  and  $\hat{i}$  by

$$\hat{S} = \frac{\sqrt{\rho_L} S_L + \sqrt{\rho_R} S_R}{\sqrt{\rho_L} + \sqrt{\rho_R}}, \qquad S = \gamma \text{ or } i, \qquad (B39a-b)$$

we obtain the following identities

$$\Delta(\rho i) = \stackrel{\sim}{\rho} \Delta i + i \Delta \rho , \qquad (B40)$$

and

$$\Delta p = \Delta \left[ \rho i (\gamma - 1) \right]$$

$$= (\hat{\gamma} - 1) \hat{\rho} \Delta i + (\hat{\gamma} - 1) \hat{i} \Delta \rho + \hat{\rho} \hat{i} \Delta \gamma , \qquad (B41)$$

so that equation (B38) yields

$$(\widehat{\gamma} - \widehat{\gamma}) \widehat{i} \Delta \rho + (\widehat{\gamma} - \widehat{\gamma}) \rho \Delta i + \widehat{\rho} (\widehat{i} - \widehat{i}) \Delta \gamma = 0 . \tag{B42}$$

The only physically acceptable solution of equation (B42) for all variations  $\Delta \rho$ ,  $\Delta i$  and  $\Delta \gamma$  is

$$\tilde{\gamma} = \hat{\gamma}$$
 and  $\tilde{i} = \hat{i}$ , (B43a-b)

given by equations (B39a-b). It now remains to determine  $\stackrel{\sim}{a}$ .

We begin by subtracting equation (B5) multiplied by  $\overset{\sim}{\text{U}}$  from equation (B10) to give

$$\frac{\widetilde{\mathbf{a}^{2}}\widetilde{\rho}\Delta\mathbf{U}}{\widetilde{\gamma}-1} = \Delta(\mathbf{U}\rho\mathbf{i}+\mathbf{U}\mathbf{p}) - \widetilde{\mathbf{U}}\Delta(\rho\mathbf{i}) - \widetilde{\mathbf{U}}\Delta\mathbf{p}$$

$$+ \Delta\left[\frac{\mathbf{U}\rho\mathbf{q}^{2}}{2}\right] - \widetilde{\mathbf{U}}\Delta\left[\frac{\rho\mathbf{q}^{2}}{2}\right] - \frac{\widetilde{\mathbf{W}}}{2}\rho\mathbf{q}^{2}\Delta\mathbf{U} , \qquad (B44)$$

which determines  $\overset{\sim}{a}$ . Simplifying equation (B44) using the following identities

$$\Delta\left[U(\rho i+p)\right] - \widetilde{U}\Delta(\rho i+p) = \widetilde{\rho} \frac{\left[\sqrt{\rho_{L}}\left(i_{L} + \frac{p_{L}}{\rho_{L}}\right) + \sqrt{\rho_{R}}\left(i_{R} + \frac{p_{R}}{\rho_{R}}\right)\right]}{\sqrt{\rho_{L}} + \sqrt{\rho_{R}}}$$
(B45)

and

we find that, after division by  $\rho\Delta U$ ,

$$\tilde{a}^2 = (\tilde{\gamma} - 1) (\tilde{H} - \frac{1}{2}\tilde{q}^2) , \qquad (B47)$$

where H is a mean enthalpy given by

$$\widetilde{H} = \frac{\sqrt{\rho_L} H_L + \sqrt{\rho_R} H_R}{\sqrt{\rho_L} + \sqrt{\rho_R}}$$
(B48)

and

$$H_{L(R)} = \frac{p_{L(R)}}{\rho_{L(R)}} + i_{L(R)} + \frac{1}{2}q_{L(R)}^{2}$$

$$= \frac{\gamma_{L(R)} p_{L(R)}}{(\gamma_{L(R)}^{-1})\rho_{L(R)}} + \frac{1}{2}q_{L(R)}^{2} . \tag{B49}$$

(N.B. If we explicitly write out equation (B47) we obtain

$$\tilde{a}^{2} = \frac{\left[\sqrt{\rho_{L}} \frac{p_{L}}{\rho_{L} i_{L}} + \sqrt{\rho_{R}} \frac{p_{R}}{\rho_{R} i_{R}}\right]}{\sqrt{\rho_{L}} + \sqrt{\rho_{R}}} \left[\frac{\sqrt{\rho_{L}}(p_{L}/\rho_{L} + i_{L}) + \sqrt{\rho_{R}}(p_{R}/\rho_{R} + i_{R})}{\sqrt{\rho_{L}} + \sqrt{\rho_{R}}} + \sqrt{\rho_{R}}\right] + \frac{\sqrt{\rho_{L}\rho_{R}}(u_{R} - u_{L})^{2}}{(\sqrt{\rho_{L}} + \sqrt{\rho_{R}})^{2}}$$
(B50)

which ensures that  $\tilde{a}^2$  is positive for real data.)

#### APPENDIX C

In this appendix we give the derivation of equations (1)-(11).

The Euler equations for compressible flows in three dimensions are

$$\mathbf{w}_{\mathsf{t}} + \mathbf{f}_{\mathsf{x}} + \mathbf{g}_{\mathsf{y}} + \mathbf{h}_{\mathsf{z}} = \mathbf{0} \tag{C1}$$

where x, y, z are Cartesian coordinates, t represents time,

$$\overset{\mathsf{w}}{\approx} = (\rho, \rho \mathbf{u}, \rho \mathbf{v}, \rho \mathbf{w}, e)^{\mathrm{T}}$$
 (C2)

$$f(w) = \left[\rho u, p + \rho u^2, \rho uv, \rho uw, u(e+p)\right]^T$$
 (C3a)

$$g(\underline{w}) = \left[\rho v, \rho v u, p + \rho v^2, \rho v w, v(e+p)\right]^T$$
 (C3b)

$$e = \rho i + \frac{1}{2}\rho(u^2 + v^2 + w^2)$$
 (C4)

and the flow variables  $\rho$ , u, v, w, e, p and i are as before. We also assume a prescribed equation of state relating p,  $\rho$  and i. Defining an invertible, twice continuously differentiable (grid) transformation from (x, y, z) space to  $(\xi, \eta, \zeta)$  space by  $x = x(\xi, \eta, \zeta)$ ,  $y = y(\xi, \eta, \zeta)$  and  $z = z(\xi, \eta, \zeta)$ , then we have

$$\frac{\partial}{\partial \bar{E}} = x_{\bar{E}} \frac{\partial}{\partial x} + y_{\bar{E}} \frac{\partial}{\partial y} + z_{\bar{E}} \frac{\partial}{\partial z} , \qquad (C5)$$

$$\frac{\partial}{\partial n} = x_n \frac{\partial}{\partial x} + y_n \frac{\partial}{\partial y} + z_n \frac{\partial}{\partial z}$$
 (C6)

and

$$\frac{\partial}{\partial \zeta} = x_{\zeta} \frac{\partial}{\partial x} + y_{\zeta} \frac{\partial}{\partial y} + z_{\zeta} \frac{\partial}{\partial z}$$
 (C7)

using the chain rule for partial derivatives. (N.B. The transformation is chosen so that the coordinate planes coincide with the body.) From equations (C5), (C6) and (C7) we find that

$$\frac{\partial}{\partial \mathbf{x}} = \frac{1}{J} \left[ J_{\xi}^{\mathbf{x}} \frac{\partial}{\partial \xi} + J_{\eta}^{\mathbf{x}} \frac{\partial}{\partial \eta} + J_{\zeta}^{\mathbf{x}} \frac{\partial}{\partial \zeta} \right] \tag{C8}$$

$$\frac{\partial}{\partial y} = \frac{1}{J} \left[ J_{\xi}^{y} \frac{\partial}{\partial \xi} + J_{\eta}^{y} \frac{\partial}{\partial \eta} + J_{\zeta}^{y} \frac{\partial}{\partial \zeta} \right] \tag{C9}$$

and

$$\frac{\partial}{\partial z} = \frac{1}{J} \left[ J_{\xi}^{z} \frac{\partial}{\partial \xi} + J_{\eta}^{z} \frac{\partial}{\partial \eta} + J_{\zeta}^{z} \frac{\partial}{\partial \zeta} \right]$$
 (C10)

where

$$J = \begin{bmatrix} x_{\xi} & y_{\xi} & z_{\xi} \\ x_{\eta} & y_{\eta} & z_{\eta} \\ x_{\zeta} & y_{\zeta} & z_{\zeta} \end{bmatrix} . \tag{C11}$$

The term  $J_{\xi}^{x}$  denotes the cofactor of  $x_{\xi}$  in the matrix given by equation (C11), i.e. generally we have

$$J_{\alpha_{1}}^{a_{1}} = \frac{\partial a_{2}}{\partial \alpha_{2}} \frac{\partial a_{3}}{\partial \alpha_{3}} - \frac{\partial a_{2}}{\partial \alpha_{3}} \frac{\partial a_{3}}{\partial \alpha_{2}}$$
 (C12)

where  $a_1, a_2, a_3 \in \{x, y, z\}$ ;  $\alpha_1, \alpha_2, \alpha_3 \in \{\xi, \eta, \zeta\}$ , with both sets considered in cyclic order. Using the expressions given by equations (C8), (C9) and (C10), equation (C1) becomes

$$J_{\mathcal{W}_{t}} + J_{\mathcal{E}}^{\mathcal{X}_{f}} + J_{\eta \eta}^{\mathcal{X}_{f}} + J_{\zeta}^{\mathcal{X}_{f}}$$

$$+ J_{\mathcal{E}}^{\mathcal{Y}_{g}} + J_{\eta \eta}^{\mathcal{Y}_{g}} + J_{\zeta}^{\mathcal{Y}_{g}}$$

$$+ J_{\mathcal{E}}^{\mathcal{X}_{h}} + J_{\eta \eta}^{\mathcal{X}_{h}} + J_{\zeta}^{\mathcal{X}_{h}} = 0 .$$
(C13)

If we note, however, that

$$\frac{\partial}{\partial \xi} \left[ J_{\xi}^{x} f_{\xi} + J_{\xi}^{y} g_{\xi} + J_{\xi}^{z} h_{\xi}^{x} \right] + \frac{\partial}{\partial \eta} \left[ J_{\eta}^{x} f_{\xi} + J_{\eta}^{y} g_{\xi} + J_{\eta}^{z} h_{\xi}^{x} \right] 
+ \frac{\partial}{\partial \zeta} \left[ J_{\zeta}^{x} f_{\xi} + J_{\zeta}^{y} g_{\xi} + J_{\zeta}^{z} h_{\xi}^{x} \right] = J_{\xi}^{x} f_{\xi} f_{\xi} + J_{\eta}^{x} f_{\eta} + J_{\zeta}^{x} f_{\zeta} f_{\zeta} 
+ J_{\xi}^{y} g_{\xi} f_{\xi} + J_{\eta}^{y} g_{\eta} + J_{\zeta}^{y} g_{\zeta} f_{\zeta} f$$

(after a large amount of cancellation), and that  $J_{\infty} = (J_{\infty})_t$ , then equations (C1)-(C4) becomes (in generalised coordinates  $\xi$ ,  $\eta$ ,  $\zeta$ )

$$(J_{\underline{w}})_t + F_{\underline{F}} + G_{\underline{n}} + H_{\underline{C}} = 0$$
, (C15)

where

$$\mathbf{w} = (\rho, \rho \mathbf{u}, \rho \mathbf{v}, \rho \mathbf{w}, \mathbf{e})^{\mathrm{T}}$$
 (C16a)

$$\tilde{E} = J_{\xi}^{x} + J_{\xi}^{y} + J_{\xi}^{z} + J_{\xi}^{z}$$

$$= \left[\rho U, J_{\xi}^{x} + \rho u U, J_{\xi}^{y} + \rho v U, J_{\xi}^{z} + \rho w U, U(e+p)\right]^{T}, \quad (C16b)$$

$$\mathcal{G} = J_{\eta}^{\mathbf{x}} + J_{\eta}^{\mathbf{y}} + J_{\eta}^{\mathbf{z}} + J_{\eta}^{\mathbf{z}}$$

$$= \left[ \rho V, J_{\eta}^{\mathbf{x}} + \rho u V, J_{\eta}^{\mathbf{y}} + \rho v V, J_{\eta}^{\mathbf{z}} + \rho w V, V(e+p) \right]^{\mathbf{T}}, \quad (C16c)$$

$$e = \rho i + \frac{1}{2}\rho(u^2 + v^2 + w^2)$$
 (C16e)

and U, V, W are given by

$$U = J_{\xi}^{x}u + J_{\xi}^{y}v + J_{\xi}^{z}w \qquad (C17a)$$

$$V = J_{\eta}^{x} u + J_{\eta}^{y} v + J_{\eta}^{z} w \qquad (C17b)$$

$$W = J_{\zeta}^{x} u + J_{\zeta}^{y} v + J_{\zeta}^{z} w , \qquad (C17c)$$

together with the equation of state

$$p = p(\rho, i) . (C18)$$